

WEIGHT AND BALANCE RECORD

Part A - Weight & Balance Maintenance Data (to be completed by a Weight and Balance Control Officer (WBCO))		Revision and Re-Issue Required INDEFINITE										
Weight and Balance Report Ref: BC/XHE/1 Centre of Gravity Position (CG) is LONGITUDINAL (delete as appropriate) AFT measured.....of datum Aircraft Longitudinal/Lateral Datum 1991.4 MM FWD OF W.L.E.		Empty Weight and Empty Weight CG Weighing or Validation dated..... 01-Nov-21 <table border="1" style="width: 100%; text-align: center;"> <tr> <td>Weight (kg)</td> <td>Arm (mm)</td> <td>Index</td> </tr> <tr> <td>802.95</td> <td>2270.1</td> <td>1822738</td> </tr> </table>	Weight (kg)	Arm (mm)	Index	802.95	2270.1	1822738				
Weight (kg)	Arm (mm)	Index										
802.95	2270.1	1822738										
Configuration: FOUR SEATS		Maximum and Minimum Empty Weight & Empty Weight CG. Revision and Re-issue by WBCO is required when calculated running totals are <table border="1" style="width: 100%; text-align: center;"> <tr> <td>Weight (kg)</td> <td>Arm (mm)</td> </tr> <tr> <td>MORE THAN</td> <td>812.95</td> </tr> <tr> <td>or</td> <td>2275.64</td> </tr> <tr> <td>LESS THAN</td> <td>792.95</td> </tr> <tr> <td></td> <td>2264.47</td> </tr> </table>	Weight (kg)	Arm (mm)	MORE THAN	812.95	or	2275.64	LESS THAN	792.95		2264.47
Weight (kg)	Arm (mm)											
MORE THAN	812.95											
or	2275.64											
LESS THAN	792.95											
	2264.47											

Part B - Record of Empty Weight and Balance Changes (the person co-ordinating maintenance shall ensure that Part B is calculated and recorded in accordance with CAO 100.7)

Date	Description of Alteration	Moment Arm from Datum (mm)	Weight and Balance Change Added (+)		Removed (-)		Running Total of Empty Weight & Empty Weight CG		
			Weight (kg)	Index	Weight (kg)	Index	Weight (kg)	Index	
01-Nov-21	BASIC EMPTY WEIGHT "INITIAL VALIDATION" PER CAO 100.7 - UNUSABLE FUEL AND FULL OIL	2657.34	366	9855.541			802.95	2270.1	1822738
02-Nov-21	Film application						802.95	2270.1	1822738

Organisation: HAWKER PACIFIC - BANKSTOWN	Aircraft Type: PIPER PA28-181	Registration: VH-XHE	Page: ONE
		Serial: 2881485	

WEIGHT AND BALANCE RECORD

LOAD DATA SHEET - PAGE 1 OF 3 - AEROPLANE WEIGHT

Aeroplane Type: PIPER PA28-181
 Registration Marking: VH-XHE Serial No: 2881485

ISSUE: ONE	DATE: 01-Nov-21	EXPIRY: INDEFINITE
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AEROPLANE WEIGHT AND CENTRE OF GRAVITY DATA

ITEM	WEIGHT (Kg)	ARM (mm aft of datum)	MOMENT (Kg.mm)	CABIN CONFIGURATION
EMPTY	802.9	2270.1	1822738	FOUR SEATS TOTAL
STANDARD CABIN CONFIGURATION				
THE FOLLOWING IMPERIAL UNITS ARE FOR USE WITH THE PILOTS HANDBOOK SECTION SIX				
EMPTY	1770.2 (lb)	89.37 (in)	158207 (in lb)	FOUR SEATS TOTAL

NOTE: The above empty weights include:-

EMPTY - unusable fuel and full oil

AeroWeigh Pty. Ltd.
 BRUCE GUSCOCK
 AUTHORIZED SIGNATORY
 LICENSE 0412 58 5531

LOAD DATA SHEET - PAGE 2 OF 3 - EQUIPMENT LIST
 This list details the items included in the empty weight shown in Page 1.

Aeroplane Type: PIPER PA28-181
 Registration Marking: VH-XHE Serial No: 2881485

ISSUE: ONE	DATE: 01-Nov-21
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- | | | | |
|--------------------------------|-------------------|--------|--|
| ENGINES/PROPELLERS | | | |
| Lycoming IO-360-B4A | 1 | | |
| Sensenich 76EM8S14 | 1 | | |
| COMPASSES | | | |
| Magnetic | 1 | | |
| Remote Indicating | 1* | | |
| THERMOMETERS | | | |
| Cyl Head Temp | 1* | | |
| Oil Temp | 1* | | |
| Outside Air Temp | 1* | | |
| INDICATORS | | | |
| Airspeed | 1* | | |
| Standby Airspeed | G5 | | |
| Direction Indicator | 1* | | |
| Exhaust Gas Temp | 1* | | |
| Hourmeters | 2 | | |
| Fuel Flow | 1* | | |
| Altitude Indicator | 1* | | |
| Standby Gyro Horizon | G5 | | |
| Tacho Non-recording | 1* | | |
| Trim Indicator | 2 | | |
| Vertical Speed | 1* | | |
| Altitude Alert | 1* | | |
| CO Detector | Guardian | | |
| RADIO EQUIPMENT (TYPE) | | | |
| ADF | 0 | | |
| Autopilot | GARMIN GFC700 | | |
| Speakers | 1 | | |
| G/slope | PART NAVS | | |
| Headsets | 1 | | |
| GPS Com | GARMIN GIA64W(x2) | | |
| Audio/Mkr | GARMIN GMA1360 | | |
| Txponder | GARMIN GTX345R | | |
| PFD | GARMIN GDU1050 | | |
| MFD | GARMIN GDU1054 | | |
| ADAHRS | GARMIN GSU75 | | |
| Magnetomit | GARMIN GMU44B | | |
| Analyzer | GARMIN GE71B | | |
| Traffic | GARMIN GTS800 | | |
| TAWS-B | GARMIN G1000 | | |
| Standby | GARMIN G5 | | |
| Magnetomit | GARMIN GMU11 | | |
| INSTRUMENTS | | | |
| Altimeters | 1* | | |
| Standby Altimeters | G5 | | |
| Volt/Ammeters | 1* | | |
| Clocks | 1* | | |
| GAUGES | | | |
| Engine Oil Pressure | 1* | | |
| Fuel Contents | 2* | | |
| Fuel Pressure | 1* | | |
| LIGHTS | | | |
| Tail Strobe | 1 | | |
| White Strobes (Tips) | 2 | | |
| Inst. Full Panel | 1 | | |
| Inst. Overhead | 2 | | |
| Landing | 2 | | |
| Map | 2 | | |
| Navigation | 4 | | |
| Cockpit Courtesy | 1 | | |
| Passenger O/Head | 2 | | |
| RESTRAINT EQUIPMENT | | | |
| Lap-sash Inertia | 4 | | |
| Inflatable Restraints | 2 | | |
| Rear Baggage Straps | 2 | | |
| ELECTRICAL EQUIPMENT | | | |
| Alternator | 1 | | |
| Batteries | 2 | | |
| Starters | 1 | | |
| External Power | 1 | | |
| MISCELLANEOUS EQUIPMENT | | | |
| Dual Controls | 1 | | |
| Fire Ext.(Portable) | 1 | | |
| Heated Pilot | 1 | | |
| Cabin Heat (Muff) | 1 | | |
| Alternate Static | 1 | | |
| Piper Vent Fan | 1 | | |
| Airconditioner | 1 | | |
| Electric Trim | 1 | | |
| Wheel Fairings | 0 | | |
| DISPOSABLE LOAD LIST | | | |
| Torch | 1 | | |
| V.S.Beacon/E.L.T. | Arte | > 1000 | |
| Tow Bar | 1 | | |

* ITEMS MARKED ARE PART OF ELECTRONIC DISPLAYS

LOAD DATA SHEET - PAGE 3 OF 3 - LOADING SYSTEM

Aeroplane Type:..... PIPER PA28-181

Registration Marking:..... VH-XHE Serial No: 2891485

ISSUE:..... ONE DATE:..... 01-Nov-21

PIPER PA28-181 LOADING SYSTEM

Registration Aircraft Type Serial Number
VH-XHE PIPER PA28-181 2891485

The following is valid only for the Empty Weight specified in Page 1 of 3 Aeroplane Weight dated, 01-Nov-21 and is based on calculations using Occupant Weights of 60 to 90 Kg each.

A...NORMAL CATEGORY OPERATIONS:-

1. OCCUPANTS:-

Load Front to Rear (i.e. Front seats first)
Load Heaviest Passenger in front row

2. BAGGAGE COMPARTMENT LIMITATIONS:-

Number of	Maximum
Occupants	Baggage
One(pilot)	68.3 Kg
Two	90.7 Kg
Three	99.0 Kg
Four	0.0 Kg

3. FUEL:-

Fuel is limited only by All Up Weight

MAXIMUM TAKE-OFF WEIGHT.....1157 Kg

B...UTILITY CATEGORY OPERATIONS:-

No limitations except Utility Category all up weight limit

MAXIMUM TAKE-OFF WEIGHT966 Kg

NOTE: In utility category operations, the baggage compartment must be empty and the rear seat unoccupied

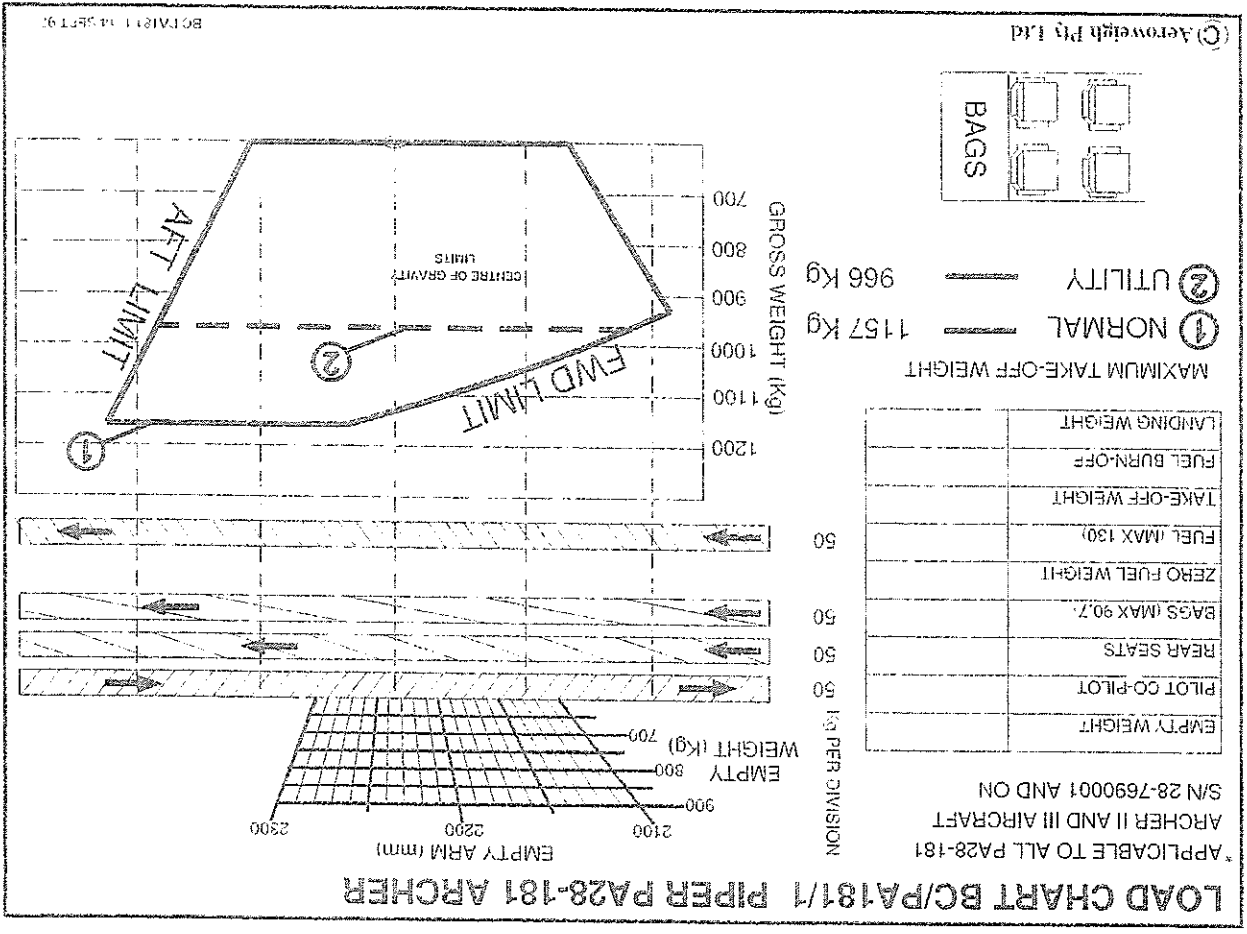
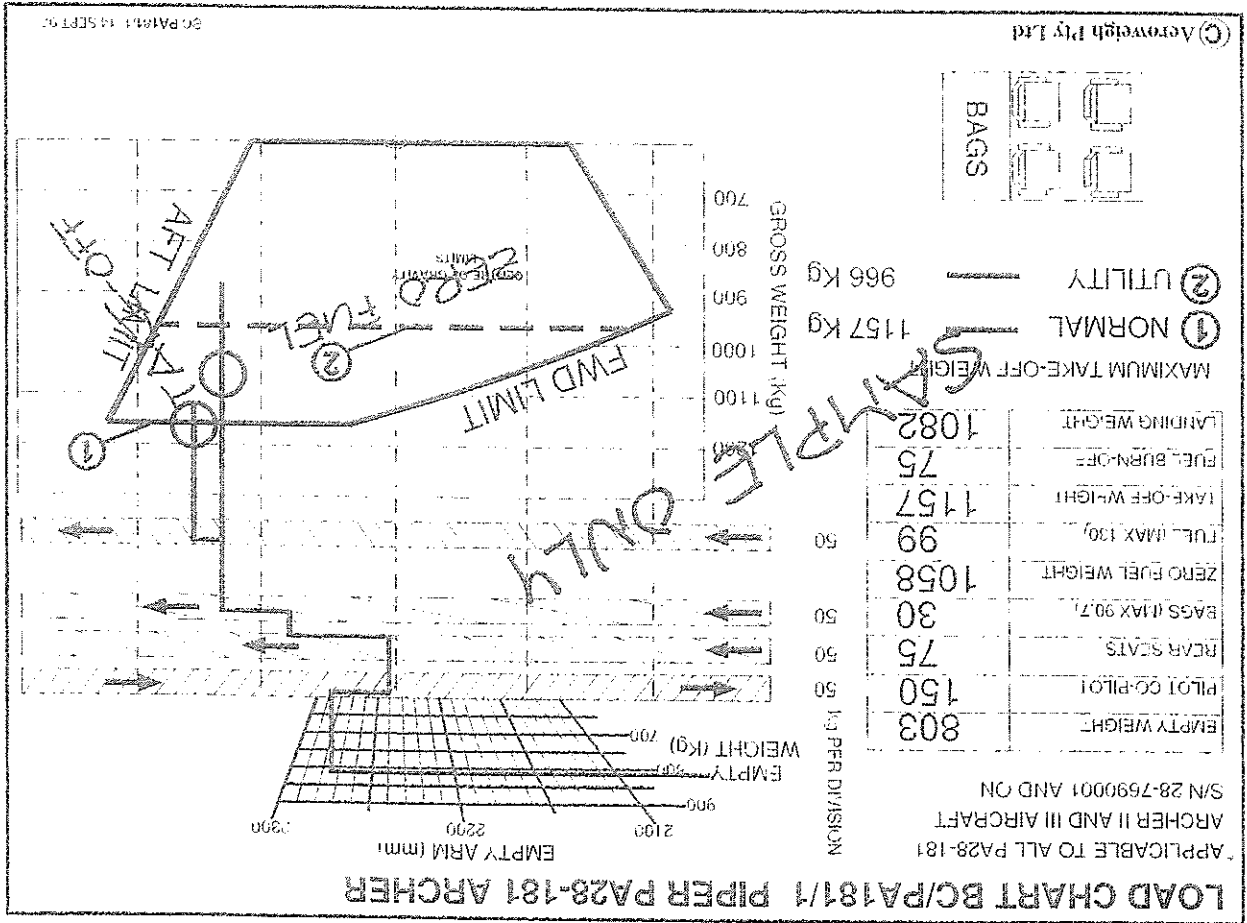
NOTE: If a full Loading Check is required, refer to Instructions and Charts in the Pilots Handbook Section Six.

INSTRUCTIONS FOR LOAD CHART BC/PA181r1
(14 SEP 1997)

PIPER PA28-181 ARCHER

1. Use standard trim sheet procedures. Enter top of chart using current Basic EMPTY Weight and Arm obtained from the Load Data Sheet.
2. Total load weights in left hand column obtaining Zero Fuel Weight and Take-off Weight.
Draw horizontal lines on the C.G. Limits Graph at the bottom of the chart.
3. From the Basic Weight Versus Arm point at the top of the chart, draw a line VERTICALLY down until it intersects one of the oblique lines on the first load item scale.
4. Move HORIZONTALLY along the load item scale in the direction of the arrow and mark a point appropriate to the load indicated in the left hand column. [e.g. with 50 Kg/Div a 100 Kg load = 2 Div.]
5. Draw a VERTICAL line down to the next load item scale and repeat the above process. (Exercise care in plotting values of segments, noting that occupant scale segments are 50 Kg)
6. Proceed VERTICALLY down the load item scales, moving to left or right as indicated by the arrows and marking scale divisions as appropriate to each load.
7. From the last cabin load item scale draw a line VERTICALLY down to intersect the Zero Fuel line previously drawn on the limits graph.
8. Continue down to the fuel scales and mark off the appropriate load, noting that FUEL has 50 Kg/Div scales, then draw a line VERTICALLY down from the last fuel point to intersect the Take off Weight line previously drawn on the limits graph.
9. The two intersection points above must not exceed the boundaries of the limits graph (heavily lined). If either point does, then the load must be rearranged and loading rechecked with steps 2 to 8.

DO NOT EXCEED MAXIMUM WEIGHTS SHOWN ON THE CHART



SECTION 6
WEIGHT AND BALANCE

PA-28-181, ARCHER III

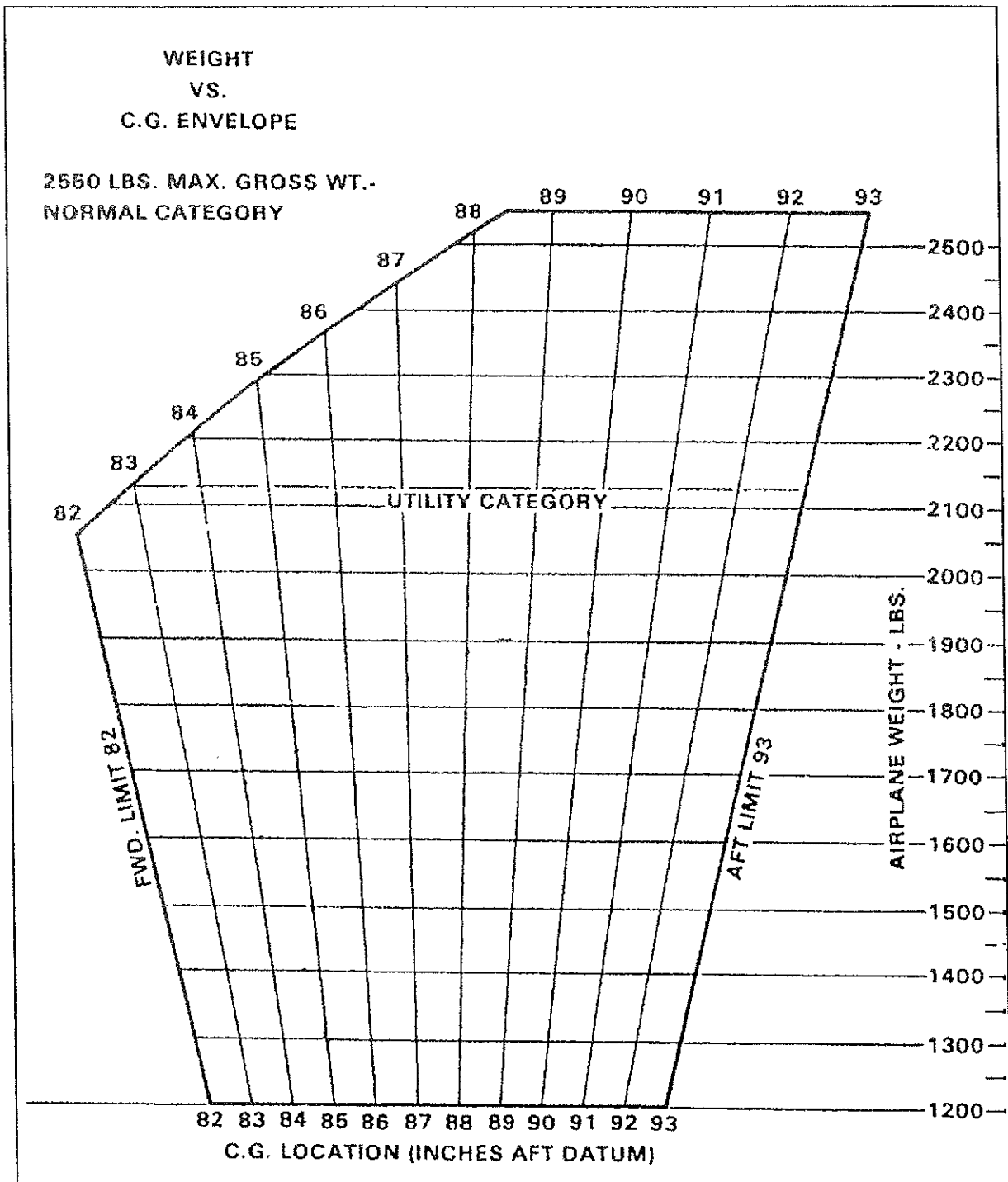
	Weight (Lbs)	Arm Aft Datum (Inches)	Moment (In-Lbs)
Basic Empty Weight			
Pilot and Front Passenger		80.5	
Passengers (Rear Seats)*		118.1	
Fuel (48 Gallon Maximum)		95.0	
Baggage (200 Lbs. Maximum)*		142.8	
Ramp Weight (2558 Lbs. Normal, 2138 Lbs. Utility Maximum)			
Fuel Allowance For Engine Start, Taxi and Run Up	-8	95.0	-760
Takeoff Weight (2550 Lbs. Normal, 2130 Lbs. Utility Maximum)			

Totals must be within approved weight and C.G. limits. It is the responsibility of the airplane owner and the pilot to ensure that the airplane is loaded properly. The Basic Empty Weight C.G. is noted on the Weight and Balance Data Form (Figure 6-5). If the airplane has been altered, refer to the Weight and Balance Record for this information.

*Utility Category Operation - No baggage or rear passengers allowed.

WEIGHT AND BALANCE LOADING FORM

Figure 6-11



C.G. RANGE AND WEIGHT

Figure 6-15

FLAPS UP TAKEOFF PERFORMANCE

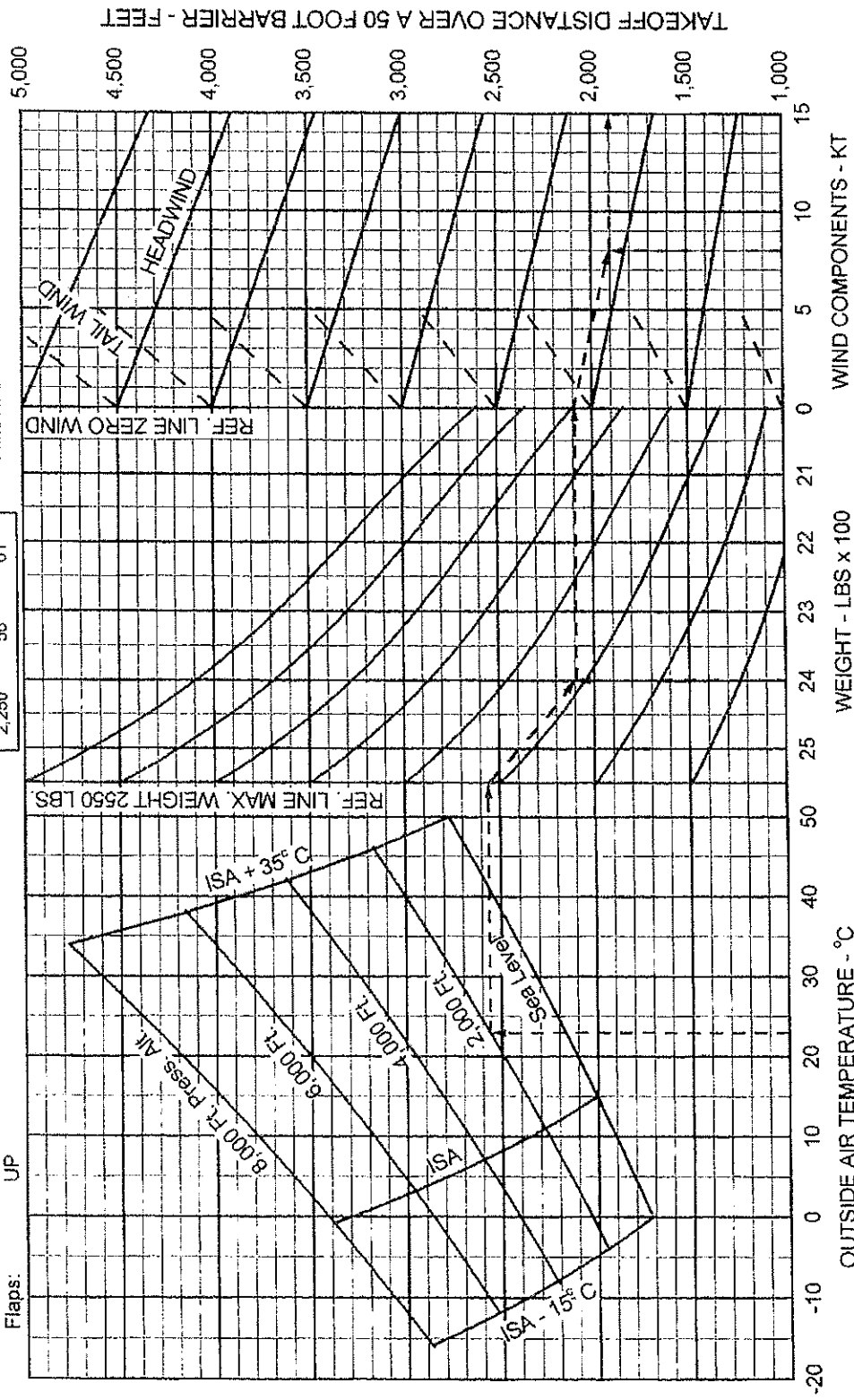
ASSOCIATED CONDITIONS:

Power: FULL THROTTLE BEFORE BRAKE RELEASE
 Air Conditioner: OFF
 Runway: PAVED, LEVEL, & DRY
 Airspeed: REFER TO TABLE AT RIGHT
 Propeller: SENSENICH 76EM8S14-0-62
 Flaps: UP

EXAMPLE:

Depart Airport Pressure Alt: 2,000 Ft
 Temperature: 23° C
 Gross Weight: 2,400 Lb.
 Headwind: 8 Kt.
 Takeoff Distance: 1907 Ft.

WT	LIFTOFF	KIAS
2,550	60	65
2,450	58	64
2,350	57	63
2,250	56	61



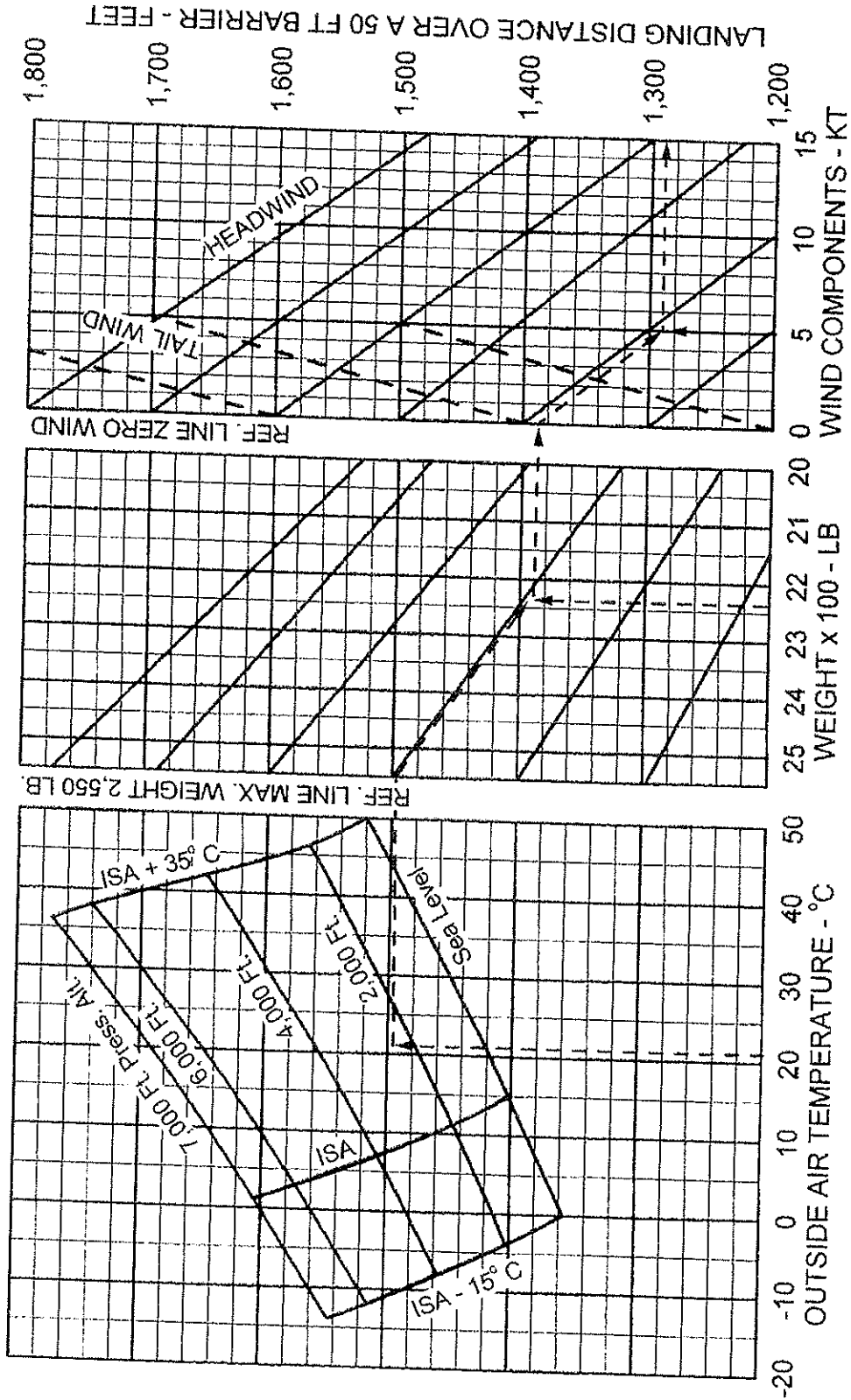
FLAPS UP TAKEOFF PERFORMANCE

Figure 5-7

LANDING PERFORMANCE
ASSOCIATED CONDITIONS

Power Off Approach, 40° Flaps, 66 KIAS, Full Stall
 Touchdown, Maximum Braking, Paved, Level, Dry Runway

EXAMPLE:
 Airport Pressure Altitude: 2,500 FT.
 O.A.T.: 21°C
 Gross Weight: 2,240 LB.
 Headwind: 5 KT.
 Landing Distance: 1,290 FT.



LANDING PERFORMANCE

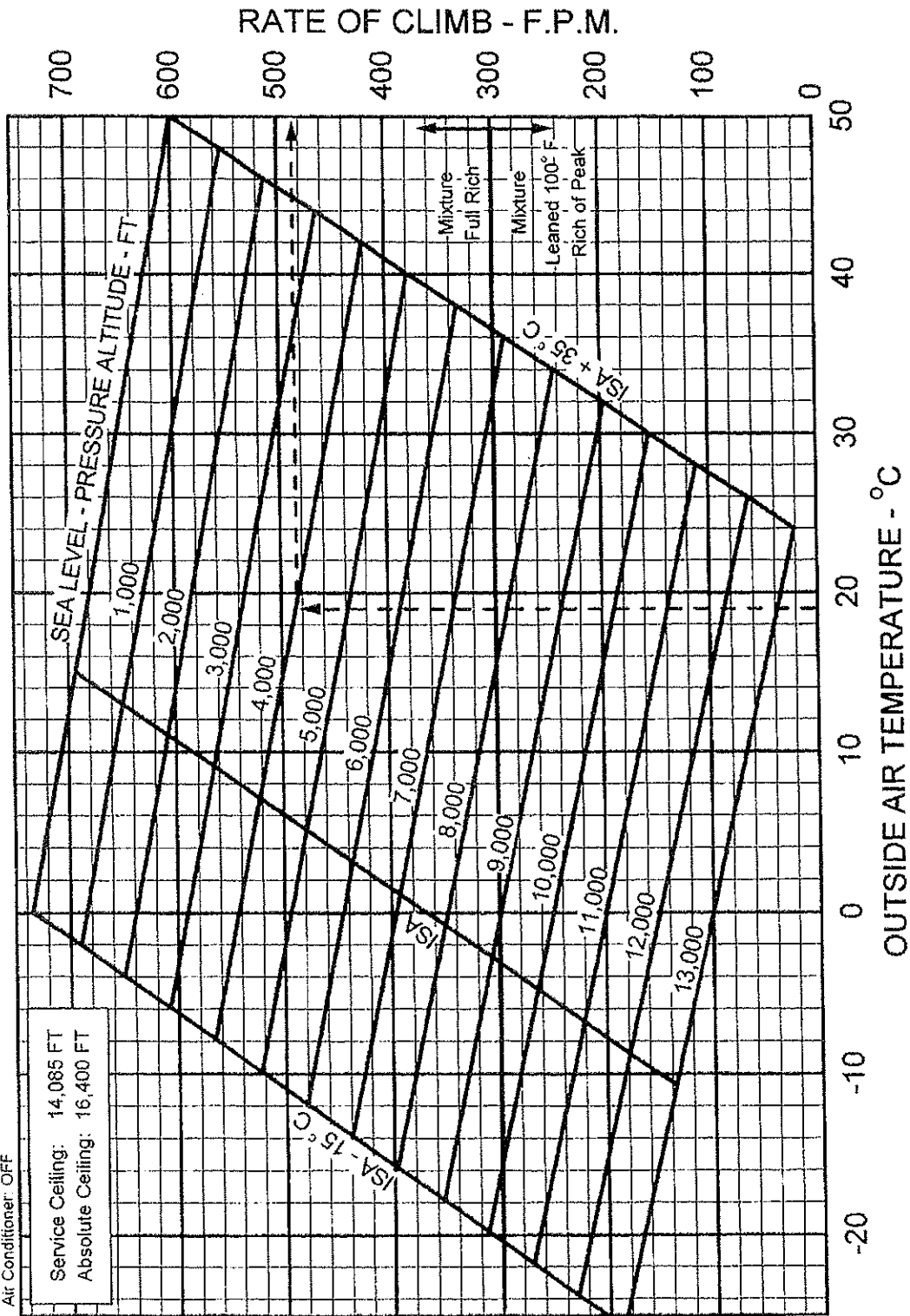
Figure 5-41

CLIMB PERFORMANCE

EXAMPLE:
Climb Pressure Alt. 4000 Ft.
Temperature: 19° C
Rate of Climb: 487 F/Min.

ASSOCIATED CONDITIONS:
Gross Weight: 2550 LBS.
Power: FULL THROTTLE
Airspeed: 76 KIAS
Flaps: UP
Air Conditioner: OFF

Service Ceiling: 14,085 FT
Absolute Ceiling: 16,400 FT



CLIMB PERFORMANCE

Figure 5-15